

The Ionospheric Challenge to Flight Safety, and the International Solution

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Flight Safety and Precision Approach

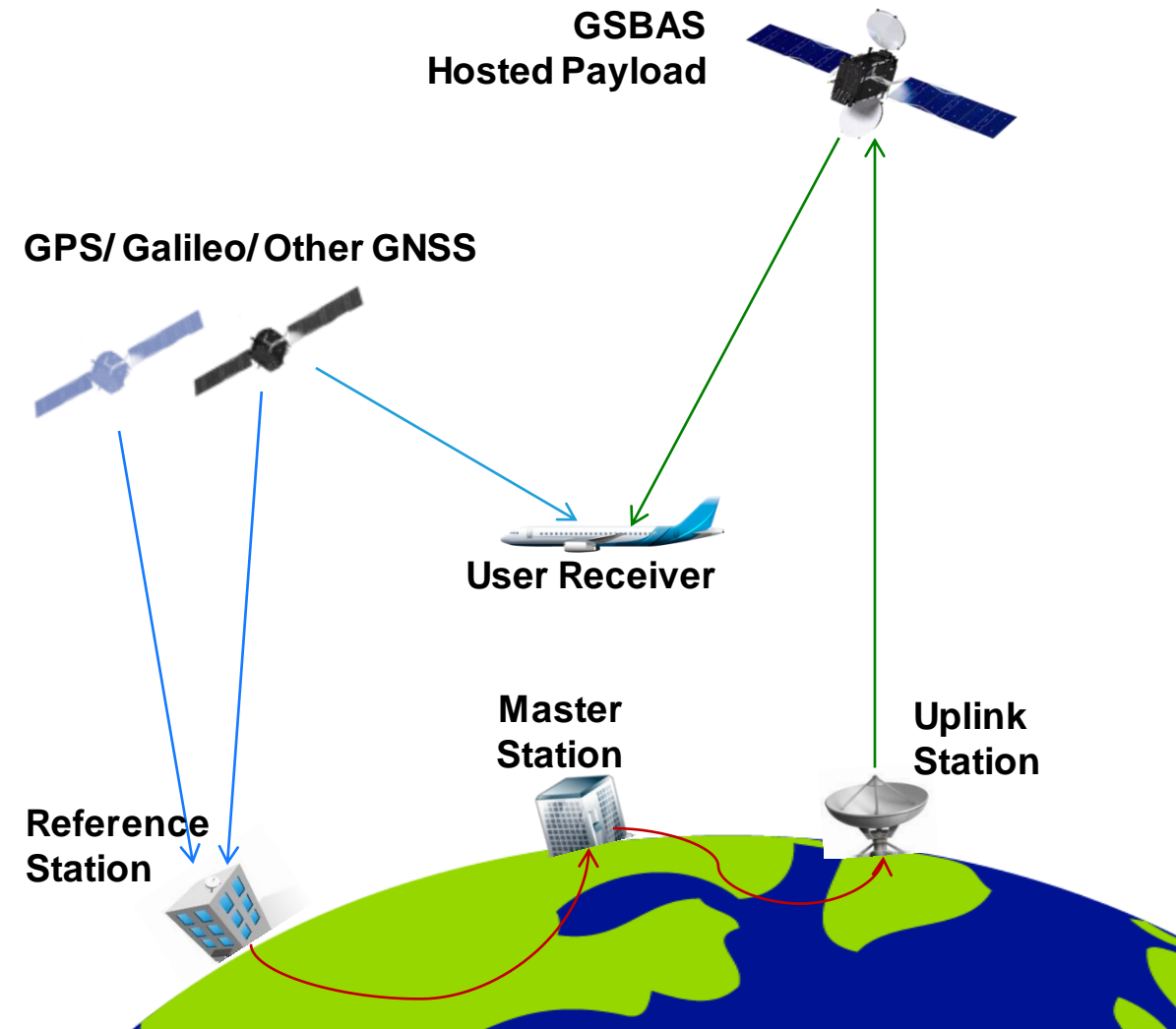


Commercial aircraft are 5 times more likely to have an accident flying a non-precision approach than flying a precision approach.

Flight Safety Foundation, March 1996



SBAS System Overview



GNSS satellites broadcast ranging signals to airborne receivers and SBAS reference stations.

Reference stations pass data to master stations, which:

- Compute ranging corrections
- Calculate 10^{-7} integrity values **for two main sources of errors**
- Satellite clock and orbit errors
- Ionosphere induced distortions

SBAS messages are sent to uplinks

SBAS messages are broadcast to hosted GEO payload

SBAS messages are re-broadcast to airborne receivers, which:

- Correct for error positions
- Calculate integrity protection bound

Entire sequence must be completed in 6 seconds to support precision approach.



ICAO ASSEMBLY RESOLUTION

- **Resolution A36 - 23**

- All approaches APV by 2016
 - Unachievable!!

← **Goal set in 2007**

← **Physics happened**

- **Resolution 37 / 11**

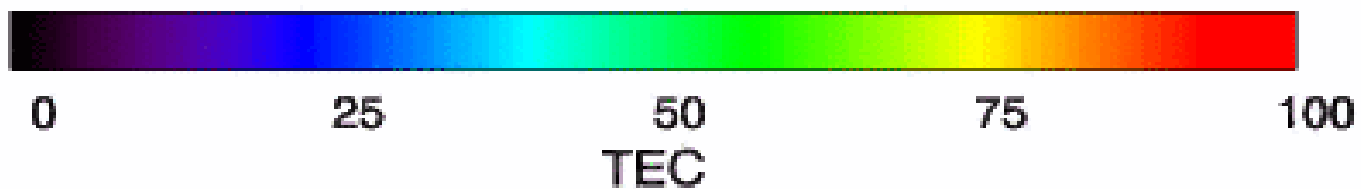
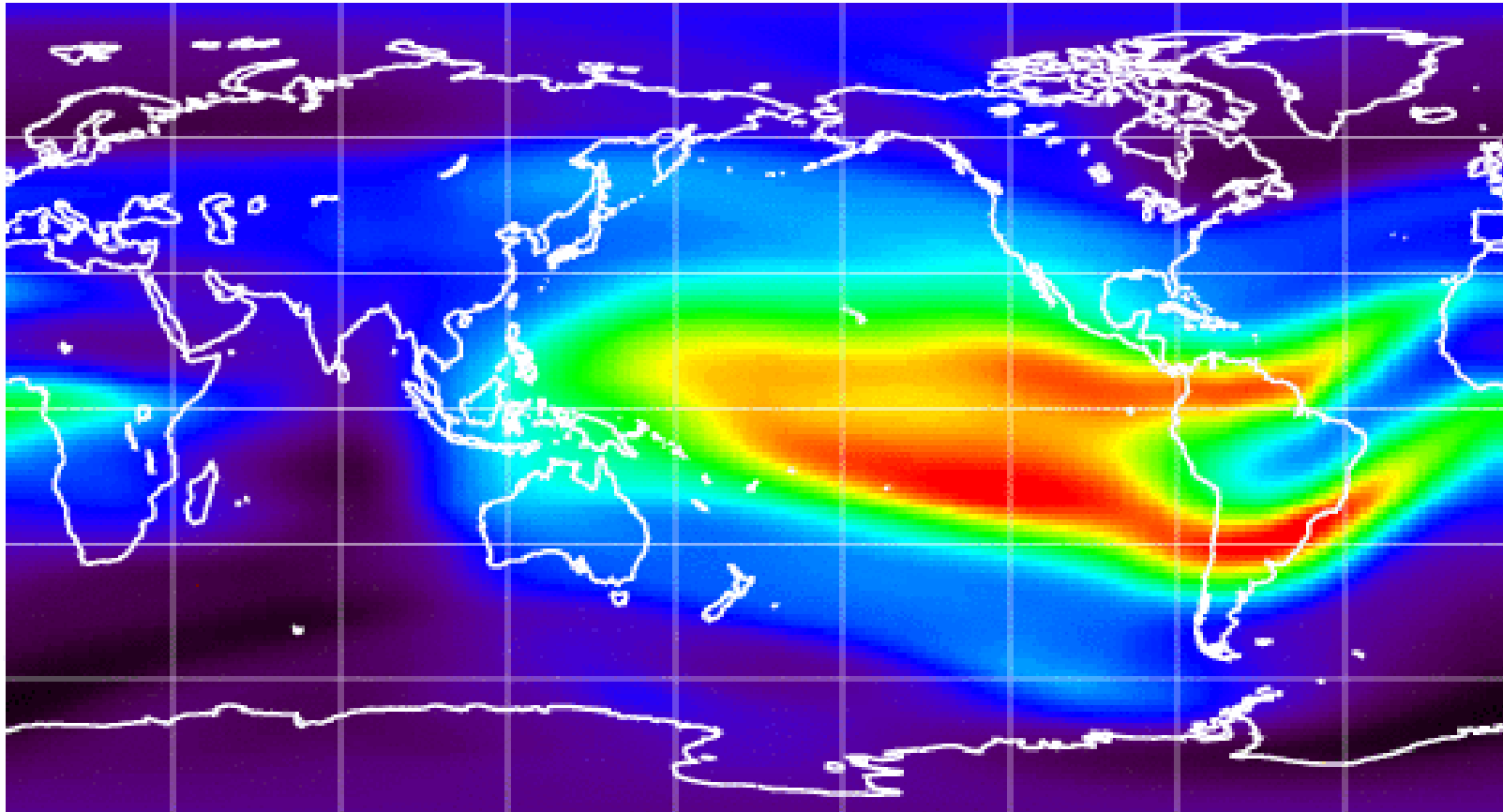
- All approaches APV by 2016
- However, if unable then straight in approaches (with limits!)

← **Goal posts moved in 2011**



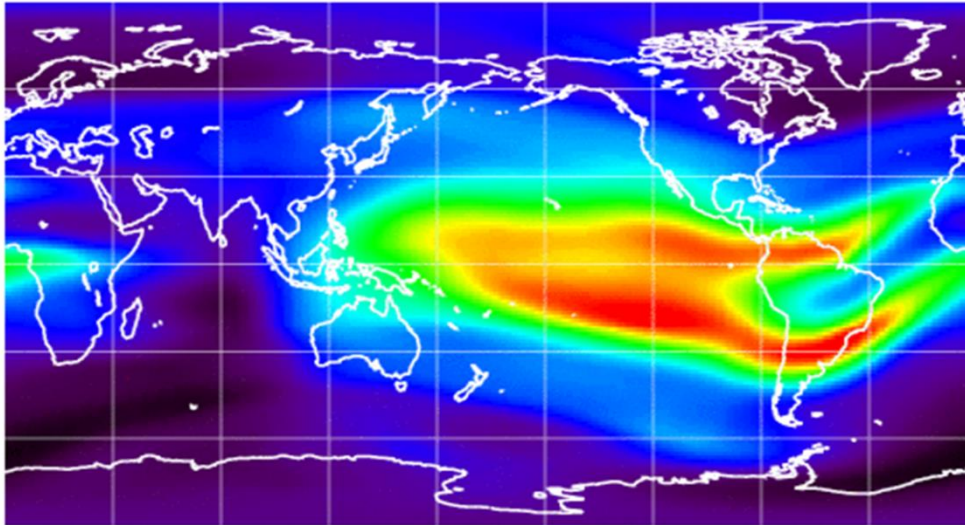


The Ionospheric Challenge





The Ionospheric Challenge

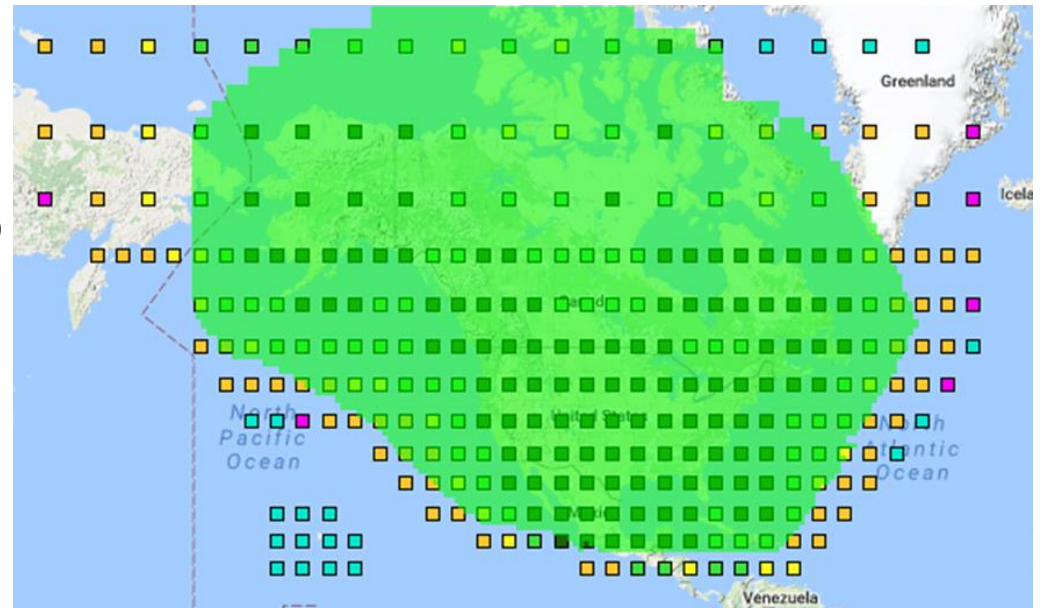


Ionosphere and GNSS errors

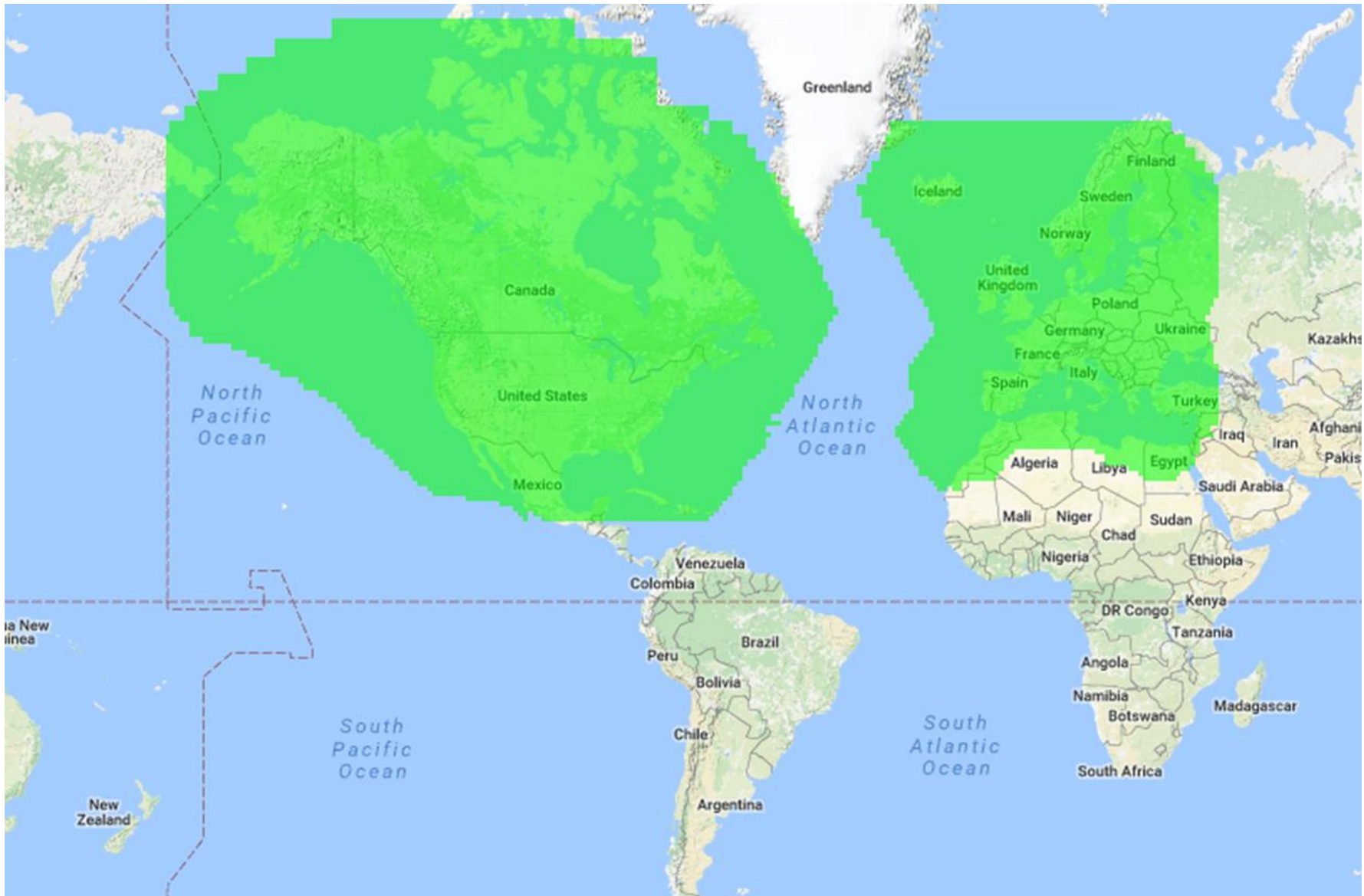
- Electron density affects signal transmission time
- Total Electron Count varies with time and location
- Solar storms can cause sudden and extreme variations
- Equatorial region is particularly affected by ionospheric changes and scintillation.

1st Generation SBAS solution

- Map of ionosphere is constructed and transmitted to user
- Vertical delay values are transmitted to user receiver
- User receiver calculates slant delays based on its location within map of vertical delays.
- Works well in mid-latitudes; very challenging in equatorial regions.



SBAS LPV Coverage



A Better Option—Fully Exploit Dual Frequencies, Multiple Constellations



L1/E1 and L5/E5a frequencies on GPS IIF, Galileo, GPS III

- 2nd frequency in protected band
- User receiver makes ionospheric corrections
- Simplifies SBAS architecture
- Solves equatorial challenge



Galileo

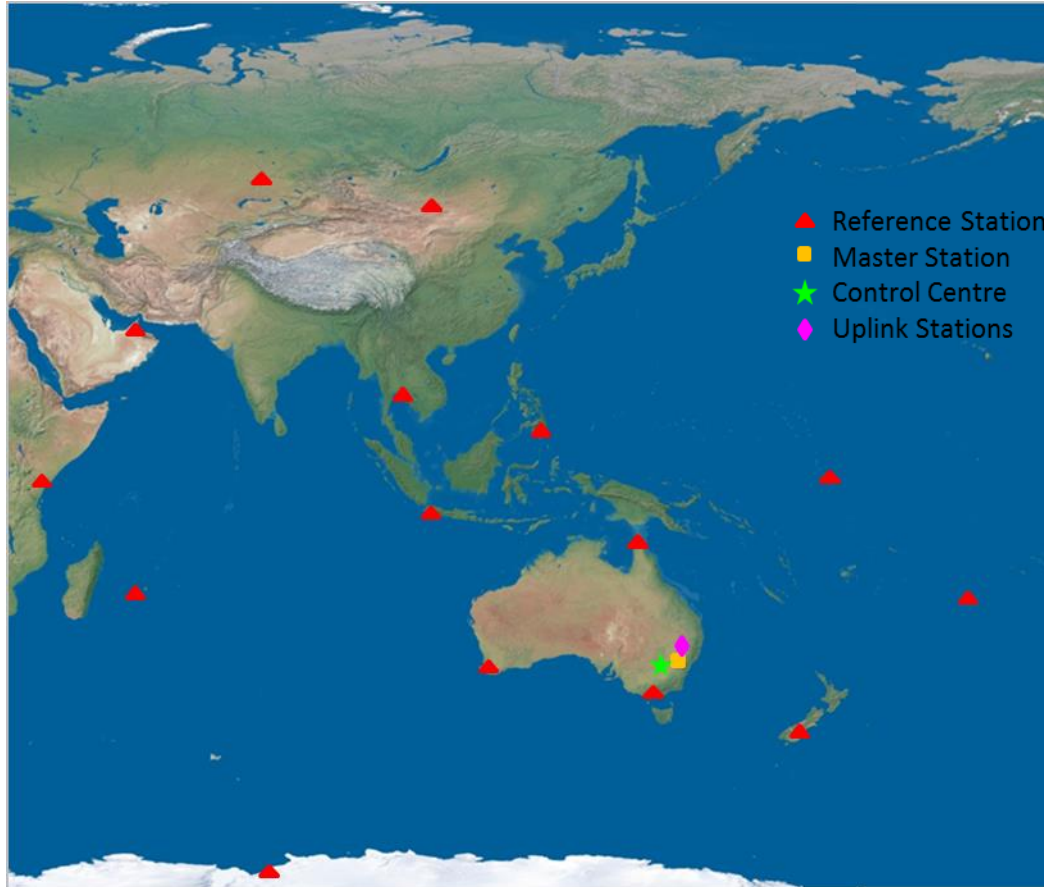


GPS III

Multiple Constellations

- More ranging signals in view
- Redundancy of GNSS providers
- Addresses sovereignty concerns

2nd Generation SBAS Testbed

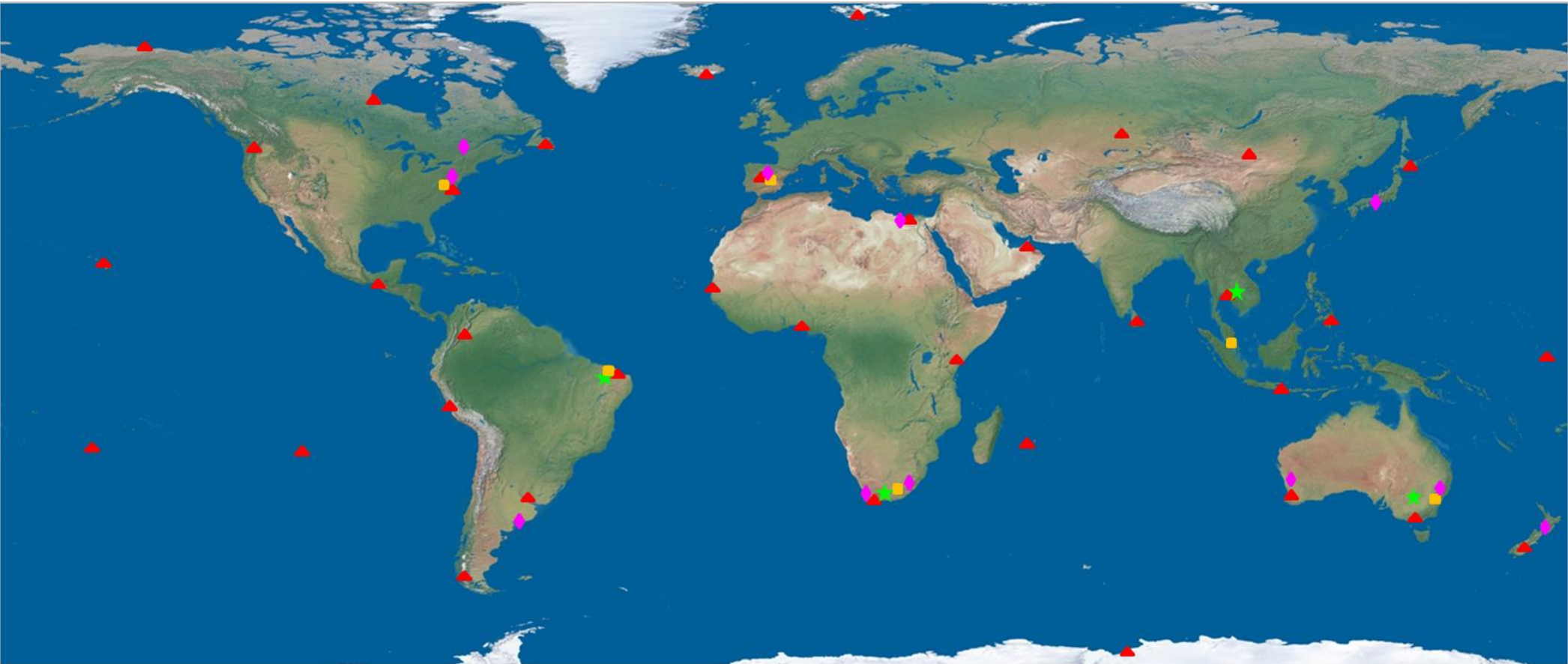


International Collaboration

- **Galileo and GPS IIF:** Open service signals on 1575.42 MHz and 1176.45 MHz
- **GMV:** Master station and control console
- **Lockheed Martin:** Uplink station and signal generator at Uralla, Australia
- **Inmarsat:** I-4F1, on orbit at 143.5° East
- **Geoscience Australia and Land Information New Zealand:**
 - Existing geodetic reference stations (CORS)
 - Coordinate with other countries for stable access to dispersed CORS data
 - Intra-systems communication links

- **Testbed architecture will use existing assets to maximum extent possible**
- **Testbed system configuration will anticipate operational system topology**

Future: Global SBAS Topology



- ▲ Reference Station
- Master Station
- ★ Control Centre
- ◆ Uplink Stations
(Sites are notional)

2nd Generation SBAS Advantages



- **Improved performance**
 - Equatorial ionospheric challenge solved by user-receiver corrections
 - Global monitoring of GNSS satellites
- **Lower cost**
- **Same architecture supports:**
 - Civil aviation
 - Emerging Safety Critical Users
 - Positive Train Control
 - Intelligent Transportation Systems
 - Maritime Navigation
 - Unmanned Aerial Vehicles
 - Automated mining

Technical approach has been validated in simulation; next step is to demonstrate with a signal-in-space testbed.

