

激光驱动微小碎片撞击效应及防护 Laser Driven Micro Flyer Hypervelocity Impact Effects on the Outer Surface of Spacecraft and Protection

Cao Yan





Team Members/ Collaborators

- Zizheng Gong: Professor, Chief Scientist, team leader. Beijing Institute of Spacecrafts Environment Engineering. China Academy of Space Technology (CAST).
- JiYun Yang: Senior Engineer, Beijing Institute of Spacecrafts Environment Engineering, CAST.
- JinChao Niu: PhD student under supervision of Prof. Zizheng gong, Beijing Institute of Spacecrafts Environment Engineering, CAST.
- Kunbo Xu: Senior Engineer, Beijing Institute of Spacecrafts Environment Engineering, CAST.
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Outline





1. Backgrounds

 Laboratory simulation of micrometeoroid/ orbital debris impacts at velocities 7km/s has long been accomplished by conventional accelerators(light-gas guns, plasma-drag accelerators)

(Expensive, low shot repetition)

 Laser Driven Flyer(LDF)
 (Compact, repeatable, relatively inexpensive)





2. Laser-driven Flyer technique

 Simulate micro-debris (diameter: ~1mm, thickness: ~ several micrometers)



Schematic diagram of LDF in CAST



Laser-driven Flyer Experiment System



2.1 Laser system

Laser beam ("top-hat" profile is desirable)

spatial profile of "top-hat" can easily produce plane shock wave, launch integrity and hypervelocity flyer, but "gauss" or exist strong spot can not obtain integrity flyer.



3D Beam profile



2.2 Flyer target

- Substrate materials: fused -silica or K9 glass
- Monolayer : AL
- Multilayer: Cr/AL

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- ♦ Thickness of flyer film: 3~10µm
- Deposition methods: magnetron sputtering, electron beam evaporation, ion beam sputtering

Glass Substrate Flyer Film Glass Substrate Ablation Layer Alz O3 Flyer Film



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2.3 Flyer Measurement





Velocity measurement (not easy: small size and low mass flyer)

- Laser profile velocity measurement system
 - Measurement error: <10% (if v<5km/s, the error is no more than 4%, and v~10km/s error ~9%)
- Scillograph (Time resolution: 2.5GHz)
- ✤ Microscope (× 50)

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Flyer velocity up to 10km/s

- K9/Cr/AI (5mm/50nm/5µm) flyer target prepared using ion beam sputtering
- A flyer plate with diameter about 1mm and 5 µm thick was accelerated to 10.4km/s at 853mJ laser energy.





3. Hypervelocity Impact Tests

*** 3.1 Optical glasses**

diameter mm	thickness μm	velocity km/s	
1	7	3.4	Single impact
1	7	4.2	Single impact
1	7	4.9	Single impact
1	7	5.5	Single impact
1.36	8	3.06	Single impact & Cumulative impacts

Impact parameter









Cumulative impacts



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a: three shots; b:one shot; c: two shots at the same position; d: three shots at the same position

Single impact: transmittance degrades $23\% \sim 45\%$ Cumulative impacts : $16\% \sim 30\%$ (v=3.06km/s)

*** 3.2 white paints (ZKS)**





Typical optical image of paints

- Solar absorptance $\alpha_{S0}=0.16$
- Hemispherical emissivity $\epsilon_{H0}=0.90\pm0.02$

white paints (ZKS)

Cumulative impacts

- *α*_s rise from 0.16 to 0.60
- Linear factor 0.023 in thickness
- Linear factor 0.244 in diameter
- $\varepsilon_{\rm H}$: no change





* 3.3 OSR (Optical Second Reflector) (α_s /ε_H : thermal control capability)

The solar absorptance of OSR rise after impact, but its emissivity without any change.







4. Protection Technique





 Parameters of DLC (Diamond-like carbon) film

- thickness: 300nm
- hardness: >30GPa
- surface roughness < 20nm</p>
- Transmitts : >80%
- Sample:
- O----OSR
- C---OSR (including DLC)



4. Protection Technique

Diamond-like carbon (DLC) films

have the unique combination of properties: high values of hardness, elastic moduli, electrical resistivity, and chemical inertness.

high values of hardness: crushed micro-debris to pieces.

The DLC films decreased the mechanical damages of surface, play an active role in protection from micro-debris.



5. Conclusion

- A laser-driven flyer ground simulation system has been developed for the launch of flyer with 1mm in diameter and a few µm in thickness to simulate space debris hypervelocity impact, achieving velocities up to 10km/s. Although the experimental results are semi-quantitative and primary, our focus is on demonstrating the suitability of laser-driven flyer technique for space debris hypervelocity impact damage, rather than drawing specific scientific conclusions from the data.
- Diamond-like carbon (DLC) films showed promise effect in protection from micro-debris impact on the outer Surface of Spacecraft.





Any questions?

